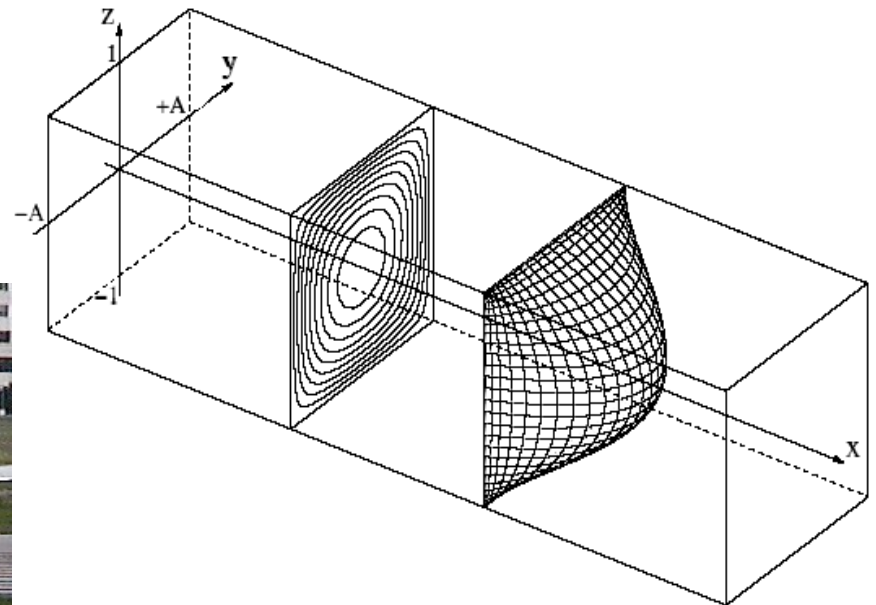




# Secondary vortices in turbulent square-duct flow

A. Bottaro, H. Soueid & B. Galletti

DIAM, Università di Genova & DIASP, Politecnico di Torino



**Goal:** hydrodynamic stability-based approach to make progress towards understanding the formation of secondary flows in turbulent ducts. Proper prediction of secondary vortices crucial in many applications.

Problem being investigated since Nikuradse (*Ph.D. Thesis*, Göttingen, 1926)

Experiments:

Brundett & Baines (1964), Gessner (1973)

Reynolds-averaged simulations:

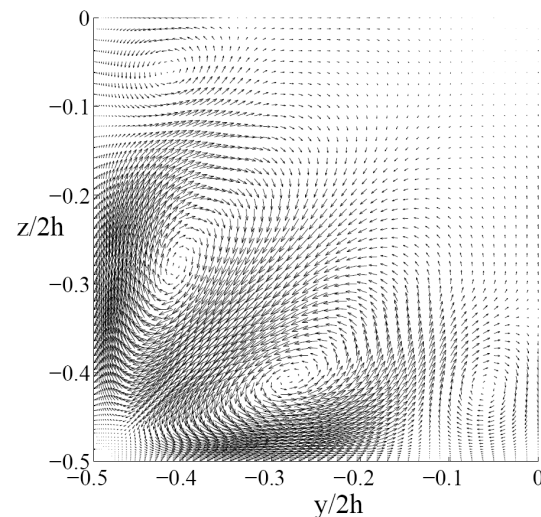
Launder & Ying (1973), Demuren & Rodi (1984)  
Mompean (1998)

DNS/LES:

Madabhushi & Vanka (1991), Gavrilakis (1992),  
Huser & Biringen (1993)

- Secondary flows near corners induced by anisotropic turbulent fluctuations
- Second-order closure underpredicts secondary vortices, possibly because of inadequate modeling of secondary shear stress components
- *“A theory of the flow structures that give rise to the observed mean flow is not yet available”*

Qualitative picture of the mean secondary flow:



# TURBULENT FLOW: AVAILABLE DNS

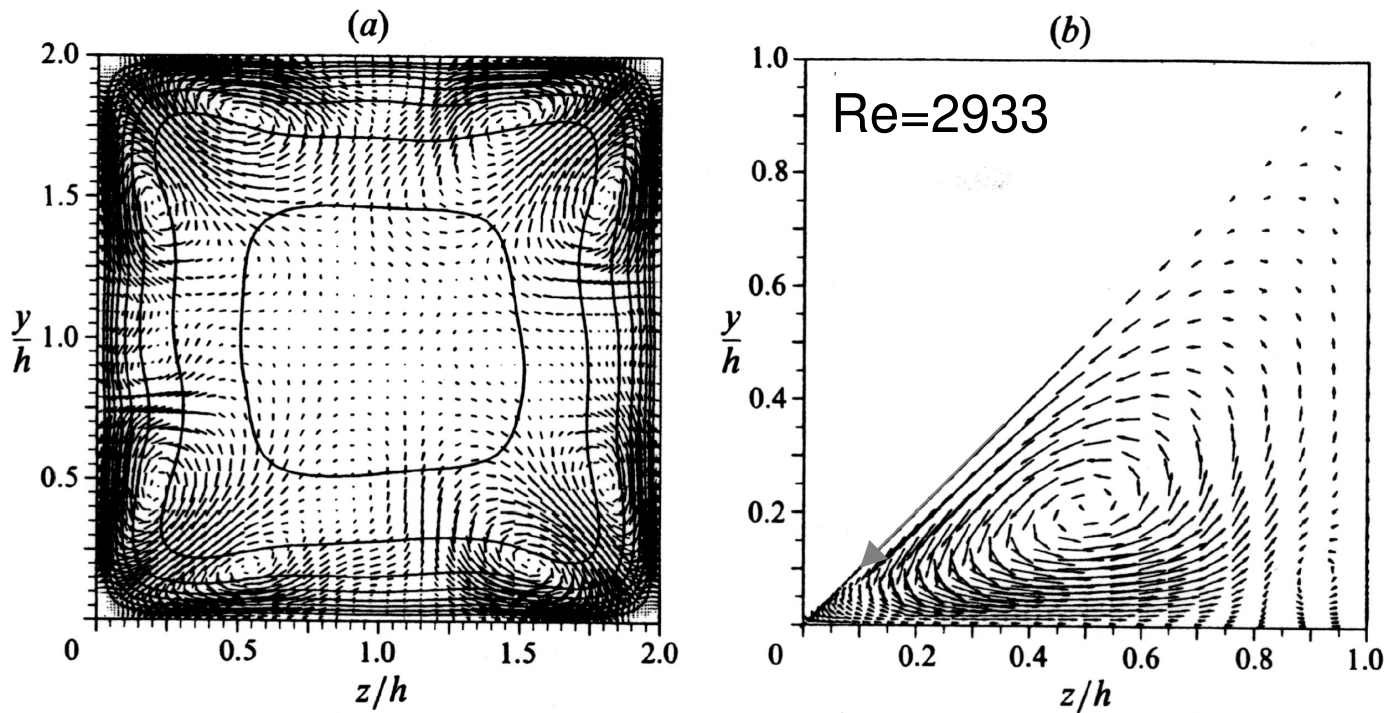


FIGURE 6. (a) Mean secondary velocity vectors and mean streamwise flow contours. The contour increment is  $4u_r$ , with the lowest value contour being nearest to the duct walls representing  $4u_r$  units. (b) Vector field in (a) averaged over all octants. Only half the vectors in each direction are shown.

# TURBULENT FLOW: AVAILABLE DNS

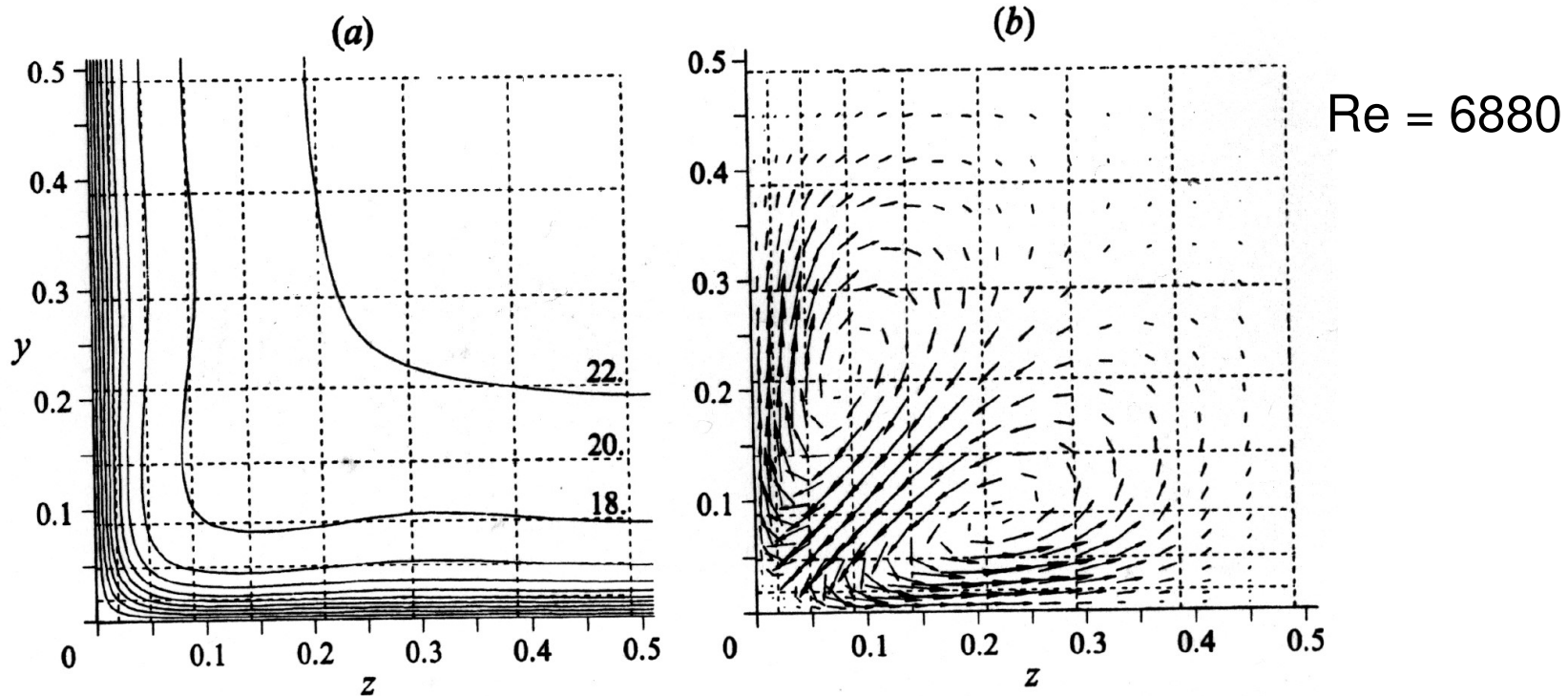
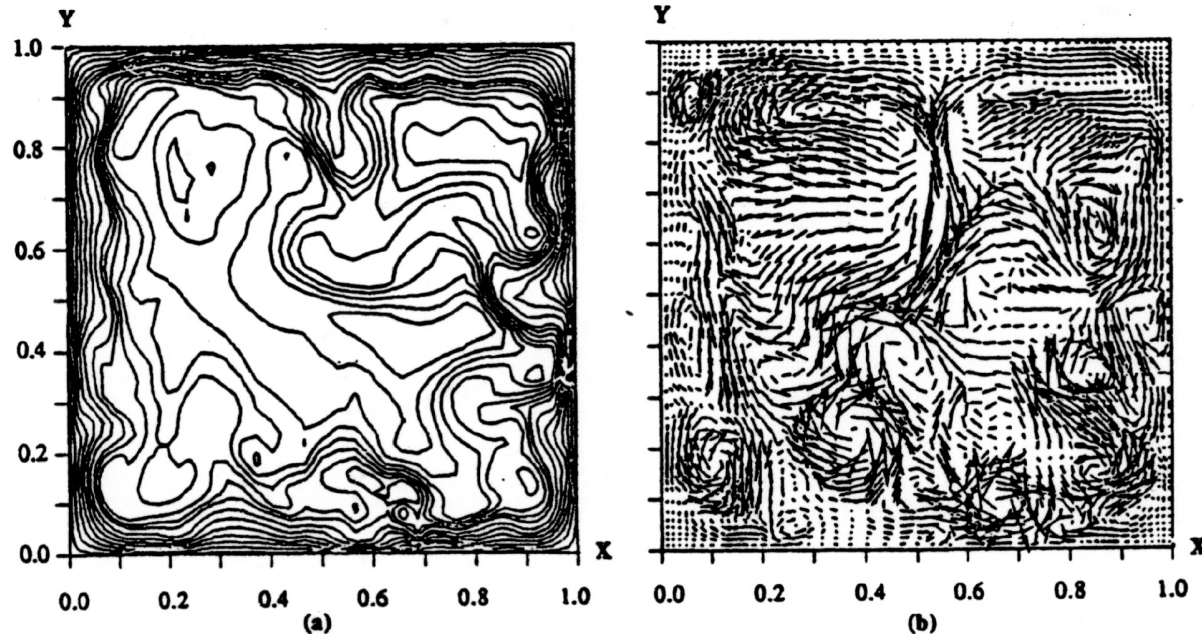


FIGURE 4. Ensemble-averaged mean velocities, Run A: (a)  $\bar{u}$ -contours, increment = 2; (b)  $\bar{v}$ ,  $\bar{w}$  velocity vectors.

A. Huser & S. Biringen, *JFM* 1993 “... the secondary flow is produced by the secondary Reynolds stresses.”

# TURBULENT FLOW: AVAILABLE LES

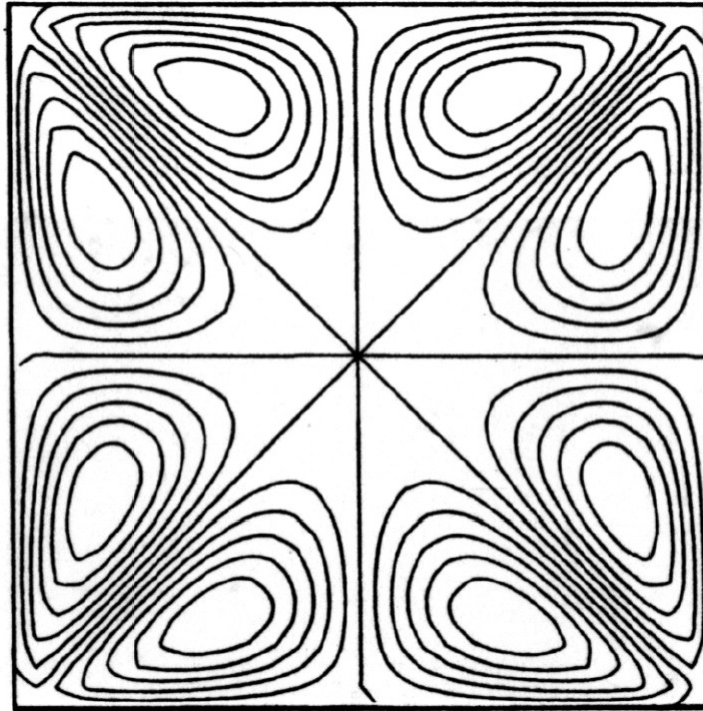


Re = 3873

FIG. 3. (a) Instantaneous streamwise velocity contours, and (b) instantaneous secondary velocity vectors in the  $Z = 0$  plane

R.K. Madabhushi & S.P. Vanka, *PoF* 1991

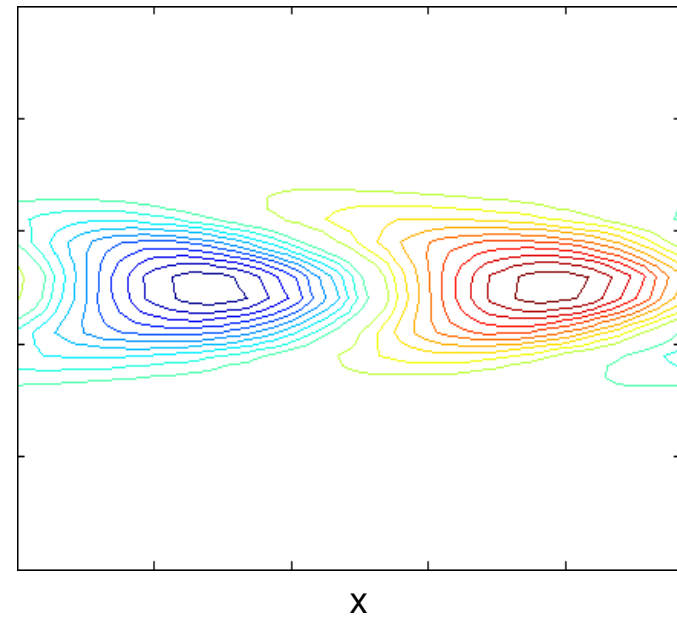
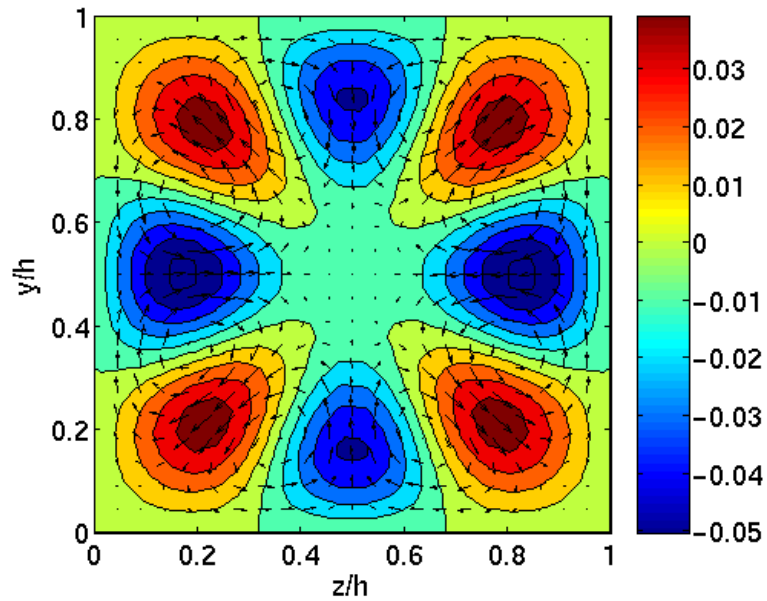
## TURBULENT FLOW: AVAILABLE RANS



**FIGURE 6.** Fully developed secondary flow streamlines in a rectangular duct obtained using the nonlinear  $K-l$  model.

C.G. Speziale, *JFM* 1987 “... linear models of turbulence can give rise to highly inaccurate predictions ...”

# “TURBULENT” FLOW: *ECS*



Isolines of the transverse velocity in a mid-plane

# OPTIMAL COHERENT STATES

Triple decomposition of the variables:

$$(u, v, w, p) = [U(y, z), 0, 0, P(x)] + [\tilde{u}(x, y, z), \tilde{v}(x, y, z), \tilde{w}(x, y, z), \tilde{p}(x, y, z)] + [u', v', w', p'](x, y, z, t)$$

with the **coherent** part  $\tilde{u}(x, y, z), \tilde{v}(x, y, z), \tilde{w}(x, y, z), \tilde{p}(x, y, z)$  “small”

$$\tilde{u}_x + \tilde{v}_y + \tilde{w}_z = 0,$$

$$U\tilde{u}_x + \tilde{v}U_y + \tilde{w}U_z = -\frac{1}{\rho} \frac{dP}{dx} - \frac{1}{\rho} \tilde{p}_x + \nu(U_{yy} + U_{zz} + \tilde{u}_{xx} + \tilde{u}_{yy} + \tilde{u}_{zz}) - (\overline{u'u'})_x - (\overline{u'v'})_y - (\overline{u'w'})_z$$

$$U\tilde{v}_x = -\frac{1}{\rho} \tilde{p}_y + \nu(\tilde{v}_{xx} + \tilde{v}_{yy} + \tilde{v}_{zz}) - (\overline{v'u'})_x - (\overline{v'v'})_y - (\overline{v'w'})_z$$

$$U\tilde{w}_x = -\frac{1}{\rho} \tilde{p}_z + \nu(\tilde{w}_{xx} + \tilde{w}_{yy} + \tilde{w}_{zz}) - (\overline{w'u'})_x - (\overline{w'v'})_y - (\overline{w'w'})_z$$

Boussinesq hypothesis:

$$-\overline{u'_i u'_j} = -\frac{P_T}{\rho} \delta_{ij} + (\overline{\nu_t} + \tilde{\nu}_t) \left[ \frac{\partial(U_i + \tilde{u}_i)}{\partial x_j} + \frac{\partial(U_j + \tilde{u}_j)}{\partial x_i} \right]$$

$\left\{ \begin{array}{l} \overline{\nu_t} \quad \text{“large”} \\ \tilde{\nu}_t \quad \text{“small”} \end{array} \right.$



Mean 1D motion	$\frac{P + P_T}{\rho U_0^2}$	$\frac{U}{U_0}$	$\frac{x}{L}$	$\frac{y, z}{h}$
----------------	------------------------------	-----------------	---------------	------------------

Leading order streamwise momentum equation:

$$0 = -\frac{1}{\rho} \frac{dP^*}{dx} + \nu(U_{yy} + U_{zz}) + (\bar{v}_t U_y)_y + (\bar{v}_t U_z)_z$$


---

Further scales to be employed for the terms left at higher order are:

Coherent motion	$\frac{x}{h/\varepsilon}$	$\frac{y, z}{h}$	$\frac{\tilde{u}}{\tilde{U}}$	$\frac{\tilde{v}, \tilde{w}}{\varepsilon \tilde{U}}$	$\frac{\tilde{p}}{\varepsilon^2 \rho \tilde{U} U_0}$
-----------------	---------------------------	------------------	-------------------------------	--	--

$\varepsilon$  not yet specified,  $\tilde{U}$  unspecified velocity scale for the coherent motion

$\tilde{U} \ll U_0, \tilde{U} \ll u_\tau$	← scale of turbulent velocity fluctuations
---	--

## Equations for the coherent motion:

$$\underbrace{\tilde{u}_x + \tilde{v}_y + \tilde{w}_z = 0,}_{\mathcal{O}(\epsilon \tilde{U}/h)}$$

$$\underbrace{U\tilde{u}_x + \tilde{v}U_y + \tilde{w}U_z}_{\mathcal{O}(\epsilon \tilde{U}U_0/h)} = \underbrace{-\tilde{p}_x/\rho}_{\mathcal{O}(\epsilon^3 \tilde{U}U_0/h)} + \underbrace{\nu \tilde{u}_{xx}}_{\mathcal{O}(\epsilon^2 \nu \tilde{U}/h^2)} + \underbrace{\nu(\tilde{u}_{yy} + \tilde{u}_{zz})}_{\mathcal{O}(\nu \tilde{U}/h^2)} + \underbrace{(\bar{\nu} \tilde{u}_x)_x}_{\mathcal{O}(\epsilon^2 \tilde{U}u_\tau/h)} + \underbrace{(\bar{\nu}_t \tilde{u}_y + \tilde{\nu}_t U_y)_y + (\bar{\nu}_t \tilde{u}_z + \tilde{\nu}_t U_z)_z}_{\mathcal{O}(\tilde{U}u_\tau/h)}$$

$$\underbrace{U\tilde{v}_x}_{\mathcal{O}(\epsilon^2 \tilde{U}U_0/h)} = \underbrace{-\tilde{p}_y/\rho}_{\mathcal{O}(\epsilon^2 \tilde{U}U_0/h)} + \underbrace{\nu \tilde{v}_{xx}}_{\mathcal{O}(\epsilon^3 \nu \tilde{U}/h^2)} + \underbrace{\nu(\tilde{v}_{yy} + \tilde{v}_{zz})}_{\mathcal{O}(\epsilon \nu \tilde{U}/h^2)} + \underbrace{(\bar{\nu}_t \tilde{u}_y + \tilde{\nu}_t U_y)_x + (2\bar{\nu}_t \tilde{v}_y)_y + [\bar{\nu}_t(\tilde{v}_z + \tilde{w}_y)]_z}_{\mathcal{O}(\epsilon \tilde{U}u_\tau/h)}$$

$$\underbrace{U\tilde{w}_x}_{\mathcal{O}(\epsilon^2 \tilde{U}U_0/h)} = \underbrace{-\tilde{p}_z/\rho}_{\mathcal{O}(\epsilon^2 \tilde{U}U_0/h)} + \underbrace{\nu \tilde{w}_{xx}}_{\mathcal{O}(\epsilon^3 \nu \tilde{U}/h^2)} + \underbrace{\nu(\tilde{w}_{yy} + \tilde{w}_{zz})}_{\mathcal{O}(\epsilon \nu \tilde{U}/h^2)} + \underbrace{(\bar{\nu}_t \tilde{u}_z + \tilde{\nu}_t U_z)_x + [\bar{\nu}_t(\tilde{v}_z + \tilde{w}_y)]_y + (2\bar{\nu}_t \tilde{w}_z)_z}_{\mathcal{O}(\epsilon \tilde{U}u_\tau/h)}$$

By imposing that the Reynolds stresses are of the same order of the convective terms (G. L. Mellor, *Int. J. Eng. Sci.* 1972) the small parameter  $\epsilon$ , that expresses The ratio of cross-stream to streamwise length scales, is found to be:

$$\boxed{\epsilon = \frac{u_\tau}{U_0}}$$

Neglecting formally small terms:

$$\left\{ \begin{array}{l} \tilde{u}_x + \tilde{v}_y + \tilde{w}_z = 0, \\ U\tilde{u}_x + \tilde{v}U_y + \tilde{w}U_z = \nu(\tilde{u}_{yy} + \tilde{u}_{zz}) + (\bar{\nu}_t\tilde{u}_y + \tilde{\nu}_tU_y)_y + (\bar{\nu}_t\tilde{u}_z + \tilde{\nu}_tU_z)_z, \\ U\tilde{v}_x = -\tilde{p}_y/\rho + \nu(\tilde{v}_{yy} + \tilde{v}_{zz}) + (\bar{\nu}_t\tilde{u}_y + \tilde{\nu}_tU_y)_x + (2\bar{\nu}_t\tilde{v}_y)_y + [\bar{\nu}_t(\tilde{v}_z + \tilde{w}_y)]_z, \\ U\tilde{w}_x = -\tilde{p}_z/\rho + \nu(\tilde{w}_{yy} + \tilde{w}_{zz}) + (\bar{\nu}_t\tilde{u}_z + \tilde{\nu}_tU_z)_x + [\bar{\nu}_t(\tilde{v}_z + \tilde{w}_y)]_y + (2\bar{\nu}_t\tilde{w}_z)_z. \end{array} \right.$$

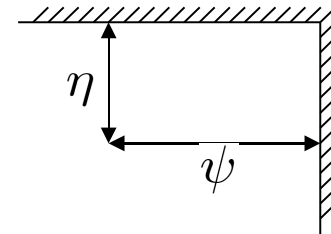
and the equations are closed by finding a suitable representation of the turbulent viscosity, e.g.:

$$\nu_t = \bar{\nu}_t + \tilde{\nu}_t = c_2 (U + \tilde{u}) l_m$$

## Mixing length:

$$l_m = 2 \frac{\eta\psi}{\eta + \psi}$$

$\left\{ \begin{array}{l} c_2 = 22 \\ l_m \end{array} \right.$  harmonic mean between the distances from two orthogonal walls



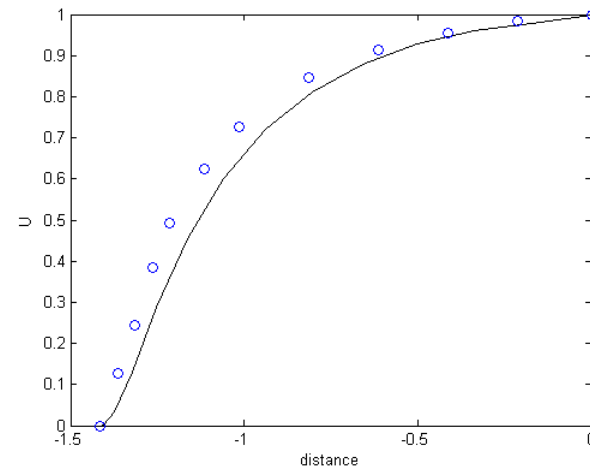
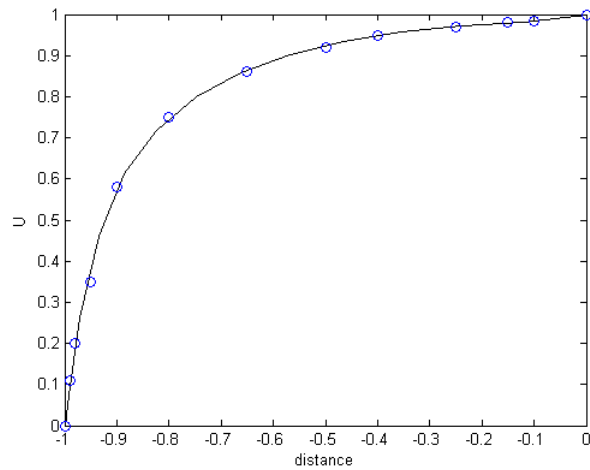
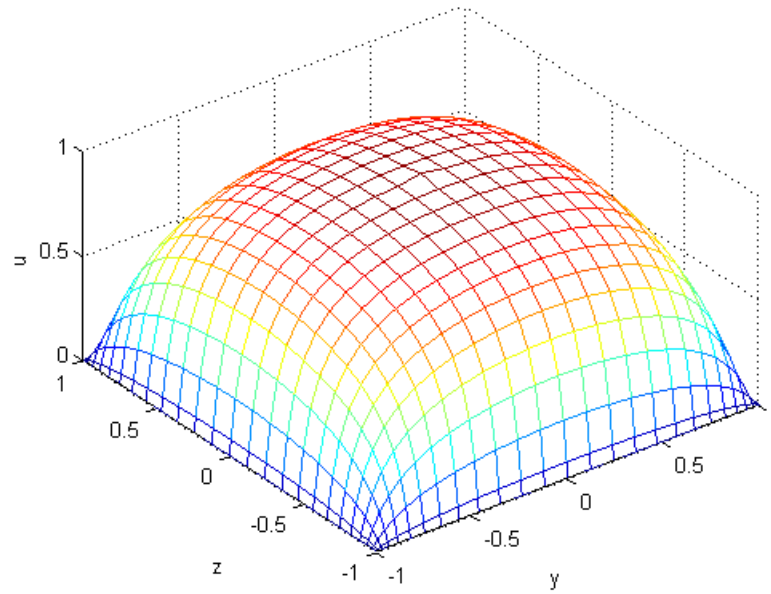
## Numerics

Collocation technique on Gauss-Lobatto grid points  $(y_i, z_j)$

$$y_i = \cos\pi(i-1)/(N-1) \text{ with } i = 1, \dots, N$$

$$z_j = \cos\pi(j-1)/(N-1) \text{ with } j = 1, \dots, N$$

$$U(y, z) = \sum_{i=1}^N \sum_{j=1}^N U_{ij} \phi_i(y) \phi_j(z), \quad \phi_i \text{ and } \phi_j \text{ Lagrangian interpolating polynomials}$$



Computed mean 1D flow and comparisons with the DNS data of Gavrilakis (1992)

## Numerics (the coherent motion)

The direct equation reads:  $\mathbf{Q}\mathbf{q}_x = \mathbf{R}\mathbf{q}$  with  $\mathbf{q} = [\mathbf{p}, \mathbf{u}, \mathbf{v}, \mathbf{w}]^T$

and upon  $x$ -discretization a recursive system is found:

$$\mathbf{q}_0 = [\mathbf{0}, \mathbf{0}, \mathbf{v}(0), \mathbf{w}(0)]^T, \quad \mathbf{q}_1 = \mathbf{G}_1 \mathbf{q}_0,$$

$$\mathbf{q}_{n+1} = \mathbf{G}_2(4\mathbf{q}_n - \mathbf{q}_{n-1}), \quad n = 1, \dots, N_L - 1$$

$\mathbf{q}_n = \mathbf{q}(n\Delta x)$ ,  $\mathbf{G}_1 = (\mathbf{Q} - \mathbf{R}\Delta x)^{-1}\mathbf{Q}$ ,  $\mathbf{G}_2 = (3\mathbf{Q} - 2\mathbf{R}\Delta x)^{-1}\mathbf{Q}$ , and  $L = N_L\Delta x$

solved with *Singular Value Decomposition*.

Constraint:  $\frac{1}{2} \int_{-1}^{-1} \int_{-1}^{-1} [\tilde{v}(0, y, z)^2 + \tilde{w}(0, y, z)^2] dy dz = E_0 = 1$

Questions:

What initial condition?  
Is there some **extremum** principle?

In “classical” stability theory it is customary to focus on the transient growth of disturbances, and to search for the initial condition that maximizes a disturbance norm (such as a kinetic energy-like norm), that reads in the present case:

$$E(x) = \frac{1}{2} \int_{-1}^{-1} \int_{-1}^{-1} \tilde{u}(x, y, z)^2 + \varepsilon^2 [\tilde{v}(x, y, z)^2 + \tilde{w}(x, y, z)^2] dy dz$$

In **turbulent flows** there are suggestions (Malkus 1956, Busse 1970, Plasting & Kerswell 2005) that statistical extreme states are reached, related to the degree of disorganization (entropy) of the motion.

It might thus be a sensible thing to maximize the **rate of viscous dissipation** ...

$$2\overline{s'_{ij}s'_{ij}}$$

with  $s'_{ij} = (\frac{\partial u'_i}{\partial x_j} + \frac{\partial u'_j}{\partial x_i})/2$  the fluctuating rate of strain

In conditions of mechanical-energy equilibrium we can consider **production** instead of dissipation:

$$\mathcal{P}(x) = - \int_{-1}^{-1} \int_{-1}^{-1} \overline{u'_i u'_j} S_{ij} dy dz$$

Which at order  $\varepsilon$  reads:

$$\mathcal{P}_\varepsilon = \int_{-1}^{-1} \int_{-1}^{-1} \tilde{\nu}_t \frac{\partial U_i}{\partial x_j} \frac{\partial U_j}{\partial x_i} + \tilde{\nu}_t \left( \frac{\partial U_j}{\partial x_i} \right)^2 + \bar{\nu}_t \frac{\partial U_j}{\partial x_i} \frac{\partial \tilde{u}_i}{\partial x_j} + \bar{\nu}_t \frac{\partial U_j}{\partial x_i} \frac{\partial \tilde{u}_j}{\partial x_i} dy dz.$$

However, an appropriate (quadratic) functional might be:

$$\mathcal{I}(x) = \int_{-1}^1 \int_{-1}^1 (\tilde{u}_y^2 + \tilde{u}_z^2) dy dz$$

provided we can consider the turbulent viscosity a property of small-scale incoherent processes and can thus rule it out ...



## The optimization and the discrete adjoint system

The cost function is written in a generic way as

$$\mathcal{J} = \alpha_1 \mathcal{I}(L) + \frac{\alpha_2}{L} \int_0^L \mathcal{I}(x) dx$$

to either target the functional at the final position ( $\alpha_2 = 0$ ) or as an integral over  $x$  ( $\alpha_1 = 0$ ).

In discrete form:

$$\mathcal{J}_n = \frac{1}{2} \alpha_1 \mathbf{q}_{N_L}^T \mathbf{A} \mathbf{q}_{N_L} + \frac{\alpha_2}{2L} \sum_{n=1}^{N_L} \mathbf{q}_n^T \mathbf{A} \mathbf{q}_n \Delta x$$

with the initial constraint:  $\frac{1}{2} \mathbf{q}_0^T \mathbf{B} \mathbf{q}_0 = E_0$

## The Lagrangian functional

The constrained optimization is transformed to an unconstrained one by introducing:

$$\begin{aligned} \mathcal{L}_n = & \frac{1}{2} \alpha_1 \mathbf{q}_{N_L}^T \mathbf{A} \mathbf{q}_{N_L} + \mathbf{r}_0^T (\mathbf{q}_1 - \mathbf{G}_1 \mathbf{q}_0) + \sum_{n=1}^{N_L-1} \{ \mathbf{r}_n^T [\mathbf{q}_{n+1} - \mathbf{G}_2 (4\mathbf{q}_n - \mathbf{q}_{n-1})] + \frac{\alpha_2}{2L} \mathbf{q}_n^T \mathbf{A} \mathbf{q}_n \Delta x \} + \\ & + \frac{\alpha_2}{2L} \mathbf{q}_{N_L}^T \mathbf{A} \mathbf{q}_{N_L} \Delta x + \lambda_0 \left( \frac{1}{2} \mathbf{q}_0^T \mathbf{B} \mathbf{q}_0 - E_0 \right), \end{aligned}$$

so that an optimum is obtained when stationarity is enforced with respect to all independent variables, leading to the following discrete adjoint system to be integrated backward in space:

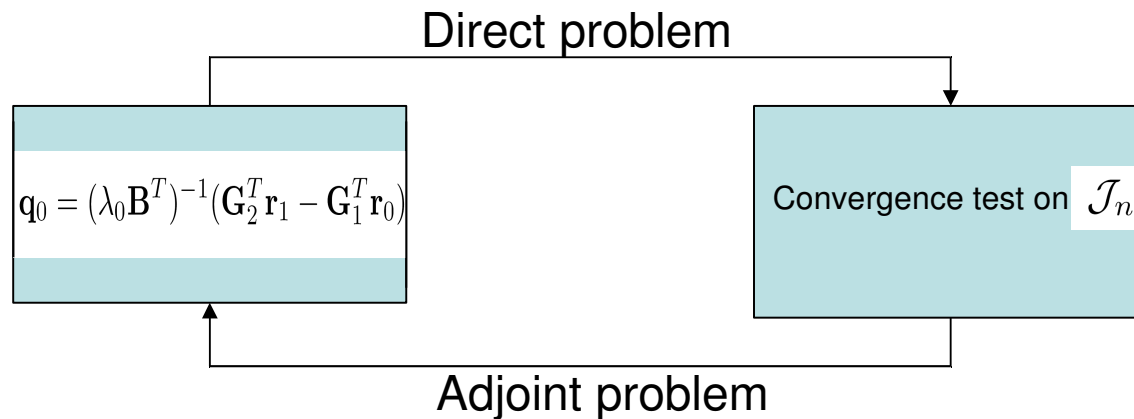
$$\mathbf{r}_{N_L} = [\mathbf{0}, \mathbf{0}, \mathbf{0}, \mathbf{0}]^T, \quad \mathbf{r}_{N_L-1} = -\left( \alpha_1 + \Delta x \frac{\alpha_2}{L} \right) \mathbf{A}^T \mathbf{q}_N,$$

$$\mathbf{r}_{n-1} = \mathbf{G}_2^T (4\mathbf{r}_n - \mathbf{r}_{n+1}) - \Delta x \frac{\alpha_2}{L} \mathbf{A}^T \mathbf{q}_n, \quad n = N_L - 1, \dots, 1$$

together with the optimality condition:

$$\mathbf{q}_0 = (\lambda_0 \mathbf{B}^T)^{-1} (\mathbf{G}_2^T \mathbf{r}_1 - \mathbf{G}_1^T \mathbf{r}_0)$$

that permits to iteratively update the inflow solution of the direct problem.



## Accuracy study for the direct problem

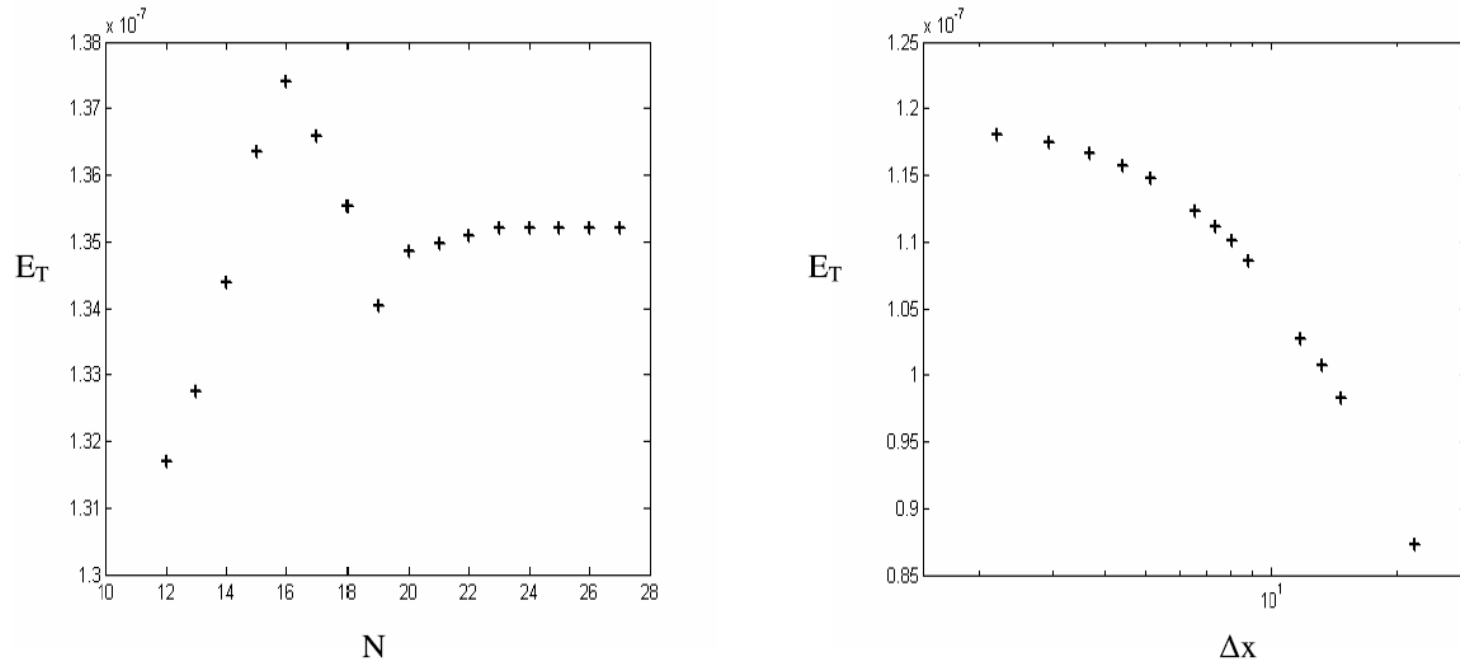
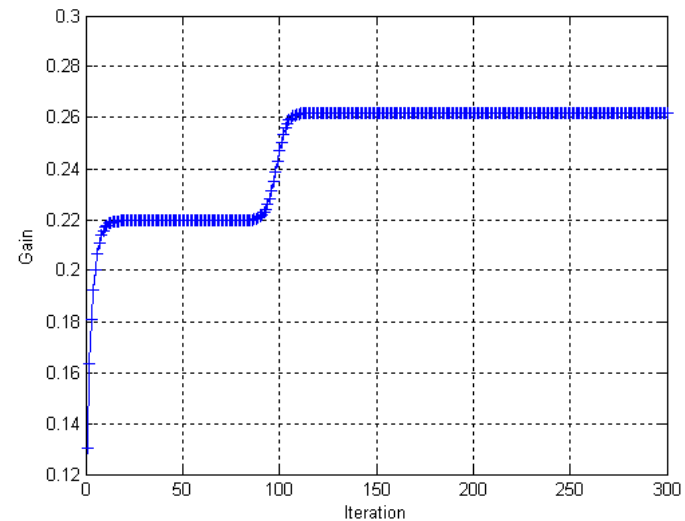
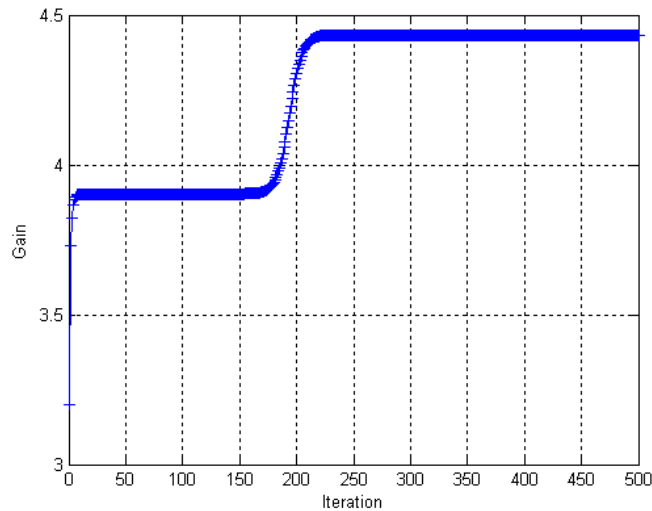


Figure 3. Grid resolution study. To obtain the figure on the left the streamwise step has been fixed at  $\Delta x = 2.2$  with  $L = 220$ ; for the figure on the right all calculations have been performed with  $N = 23$  and  $L = 396$ .

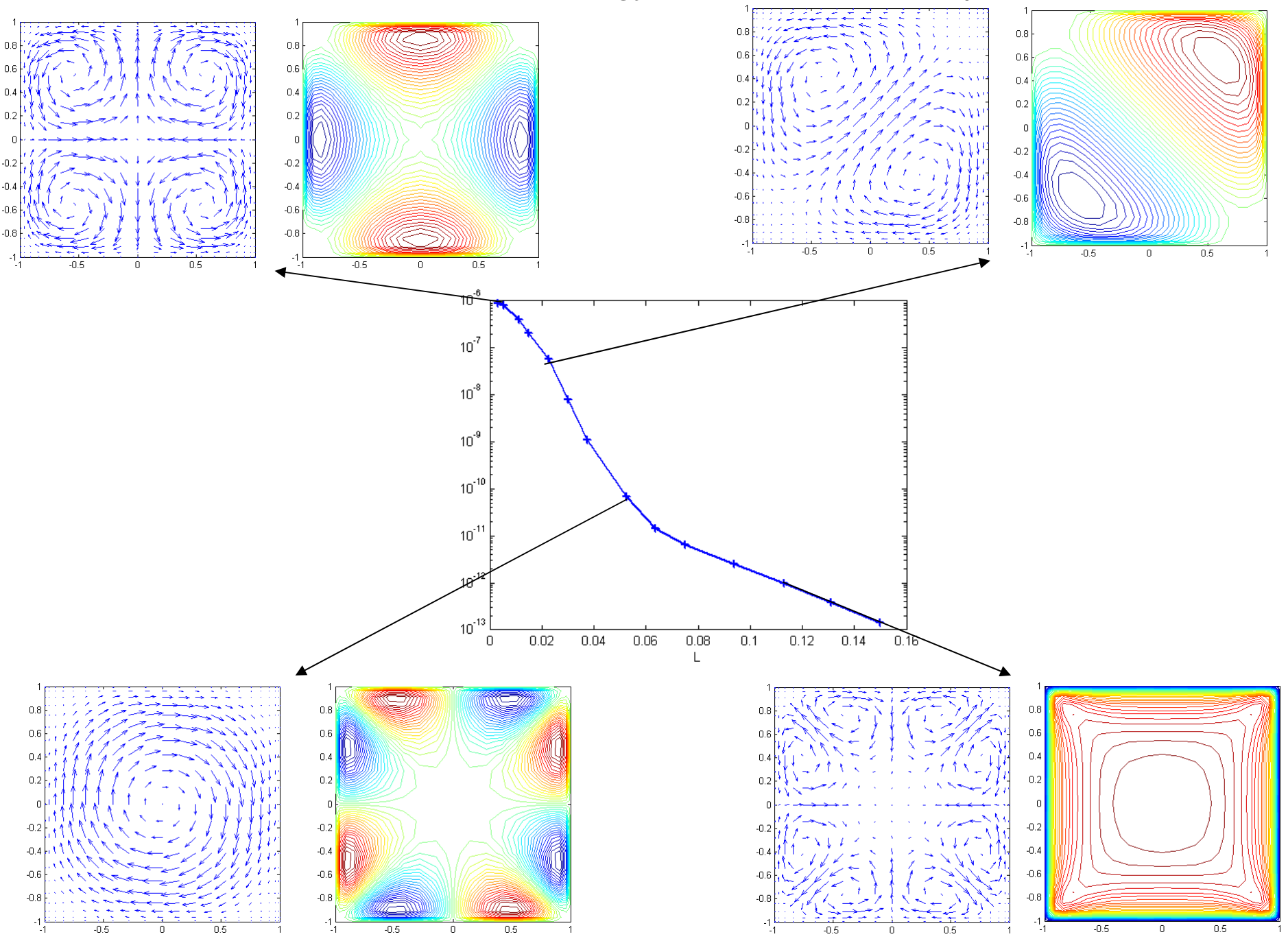
The direct-adjoint iterative procedure can be very long, pointing to the **weak selectivity** of the final “optimal” states (Luchini 2000).



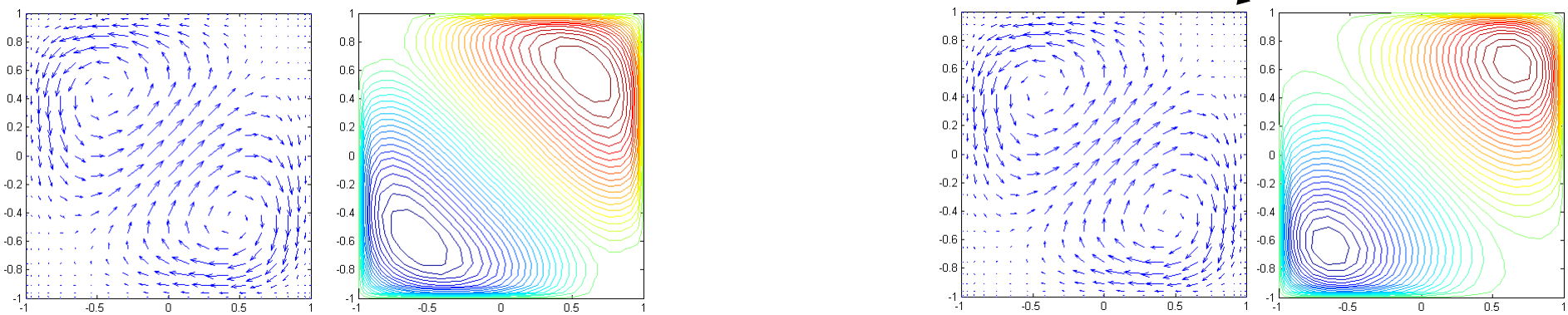
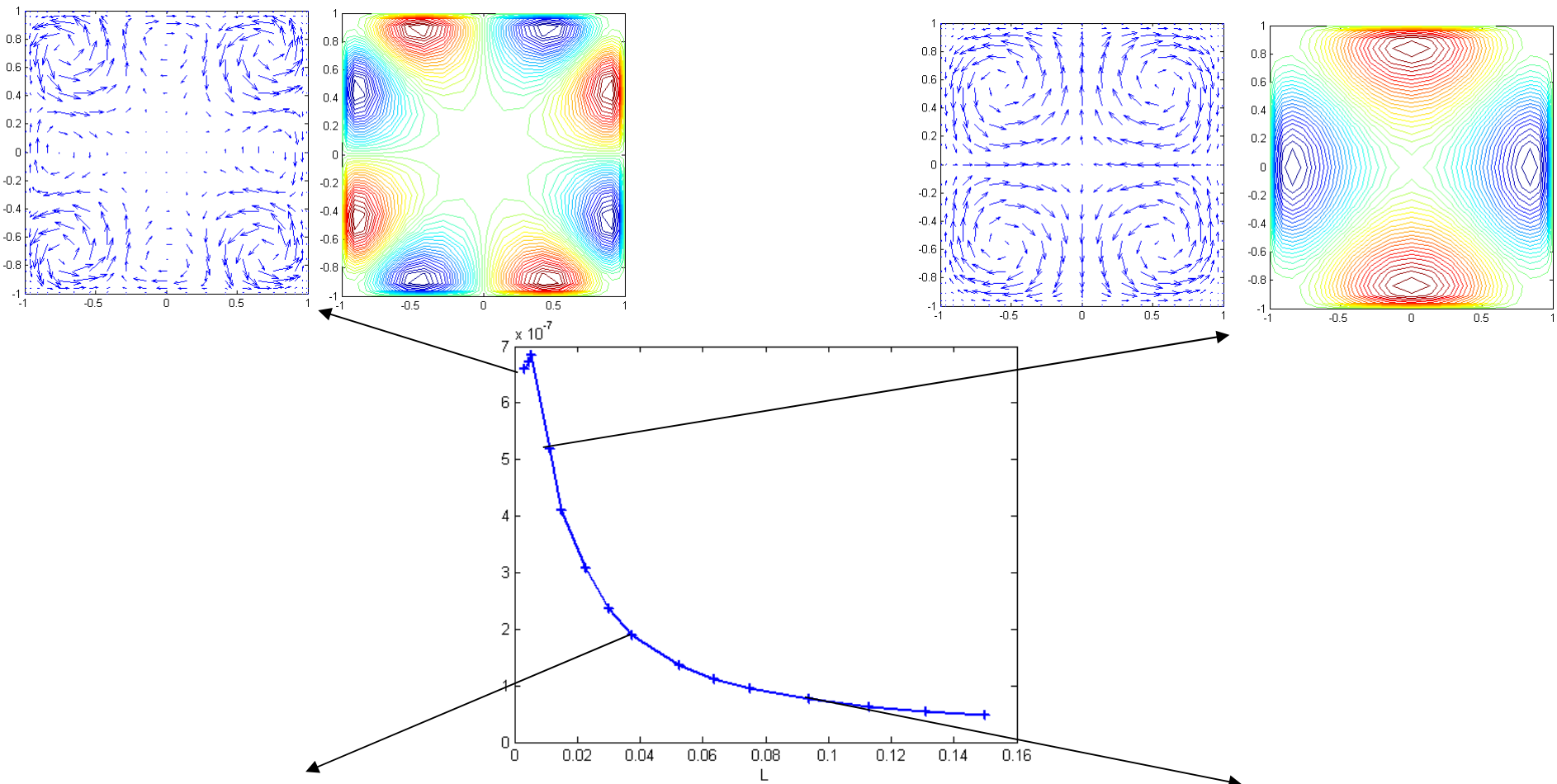
Examples of the presence of *plateaux* in the course of iterations

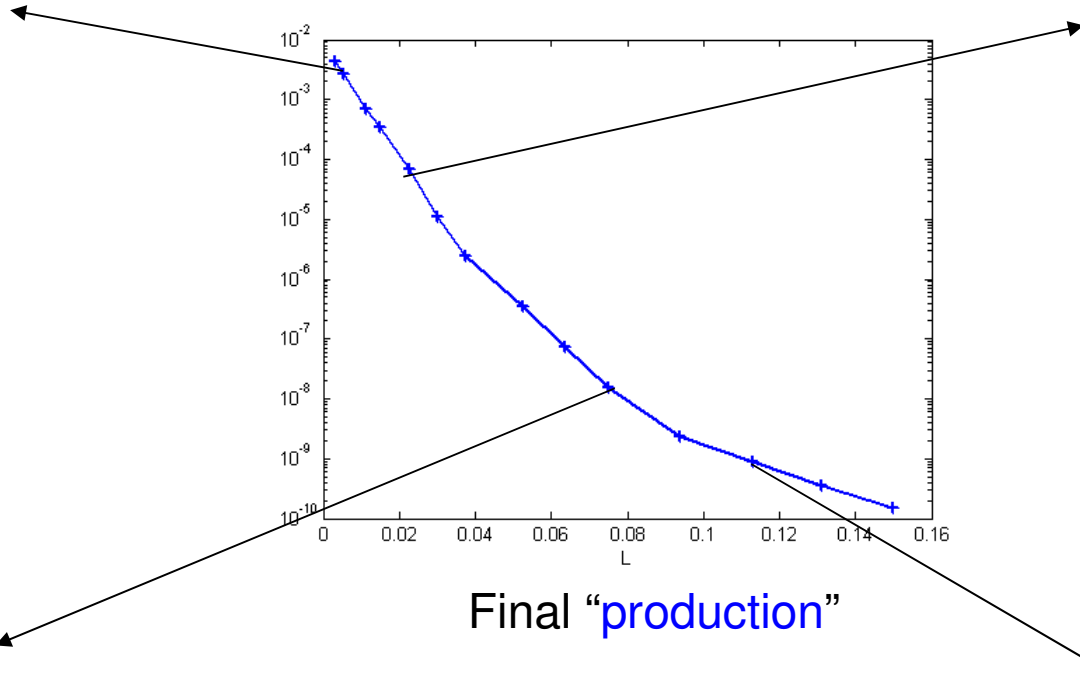
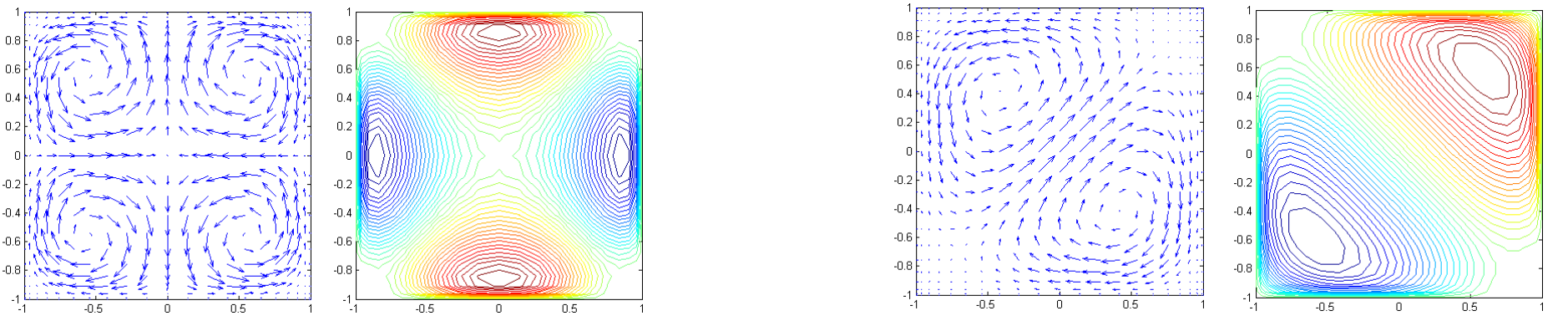
500 iterations are needed to reach convergence in short ducts (left), whereas 300 are sufficient in long ducts (right)

# Functional: kinetic energy-like norm at the final position

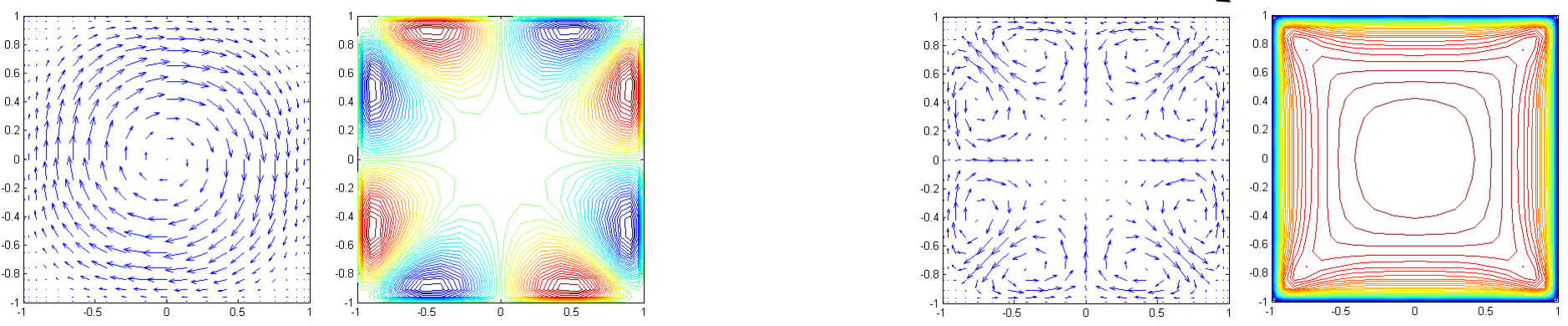


# Functional: integral over $x$ of the kinetic energy-like norm



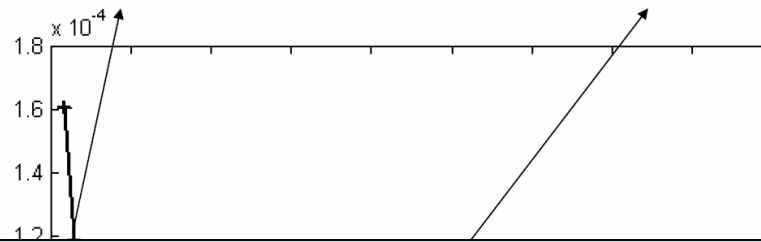
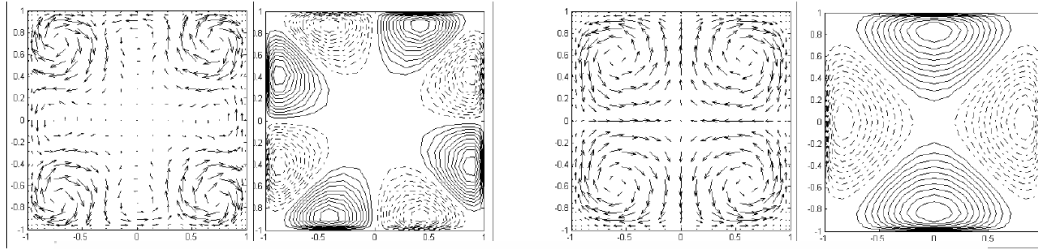


Final "production"

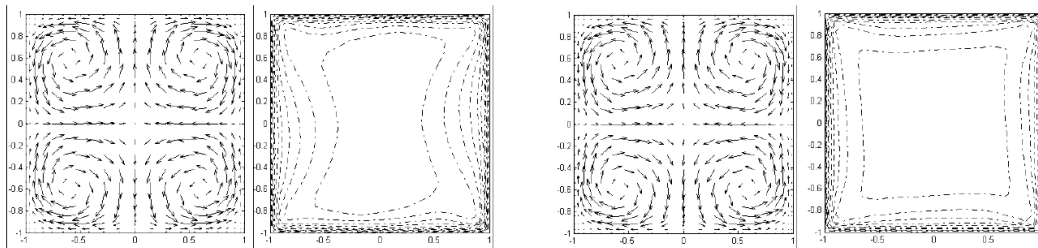
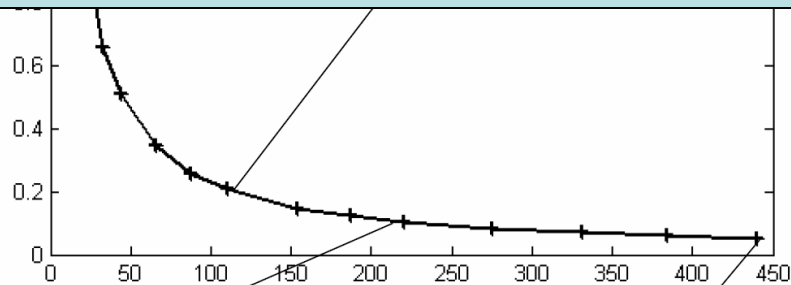




# Integral “production”



Is there an optimal length over which one should/could optimize?



## Discussion

Streamwise vorticity equation (linearized in the coherent field):

$$\begin{aligned}
 U\tilde{\omega}_x = & \nu(\tilde{\omega}_{yy} + \tilde{\omega}_{zz}) + \underbrace{U_z\tilde{v}_x - U_y\tilde{w}_x}_{P_1} + \underbrace{(\overline{v'v'} - \overline{w'w'})_{yz}}_{P_2} \\
 & + \underbrace{(\partial_{zz} - \partial_{yy})\overline{v'w'}}_{P_3} + \underbrace{\partial_x[(\overline{u'v'})_z - (\overline{u'w'})_y]}_{P_4}
 \end{aligned}$$

Nonlinear, fully developed flow:  $\tilde{v}\tilde{\omega}_y + \tilde{w}\tilde{\omega}_z = \nu(\tilde{\omega}_{yy} + \tilde{\omega}_{zz}) + P_2 + P_3$

A classical Boussinesq closure with a constant eddy viscosity for  $P_2$  and  $P_3$  would decouple this equation from the streamwise momentum equation, and yield  $\tilde{v} = \tilde{w} = 0$ , which is the reason why traditionally a nonlinear relation for the Reynolds stress tensor has been considered necessary.

**However**, here  $P_1$  (*mean shear skewing*) and  $P_4$  ( $\tilde{\omega}$  -  $\tilde{u}$  coupling)

are also present, and can trigger secondary coherent vortices through spatially transient effects.

## Outlook and conclusions

- **Turbulent square duct**: triple decomposition of the flow variables, very simple closure, coherent state of small amplitude
- Technique borrowed from optimal control, of common use in stability theory: direct-adjoint optimization to search for cross-stream (coherent) vortices capable of maximizing certain functionals
- One-dimensional turbulent-like mean flow  $U(y,z)$  can sustain the growth of secondary motion of a cross-stream scale comparable to that found in experiments
- Nonlinearities (not considered here) are responsible for *maintaining* the coherent states
- Variety of structures present and weak selectivity of the “optimal” states; the functional is the largest in the limit of very small duct lengths, *suggestive* pictures are found for ducts sufficiently long
- No need for anisotropic turbulent viscosity when a streamwise inhomogeneous approach is taken (*i.e.* when vortices are considered to arise out of transients)
  
- No clear indications on the existence of extremum principles in turbulent flow

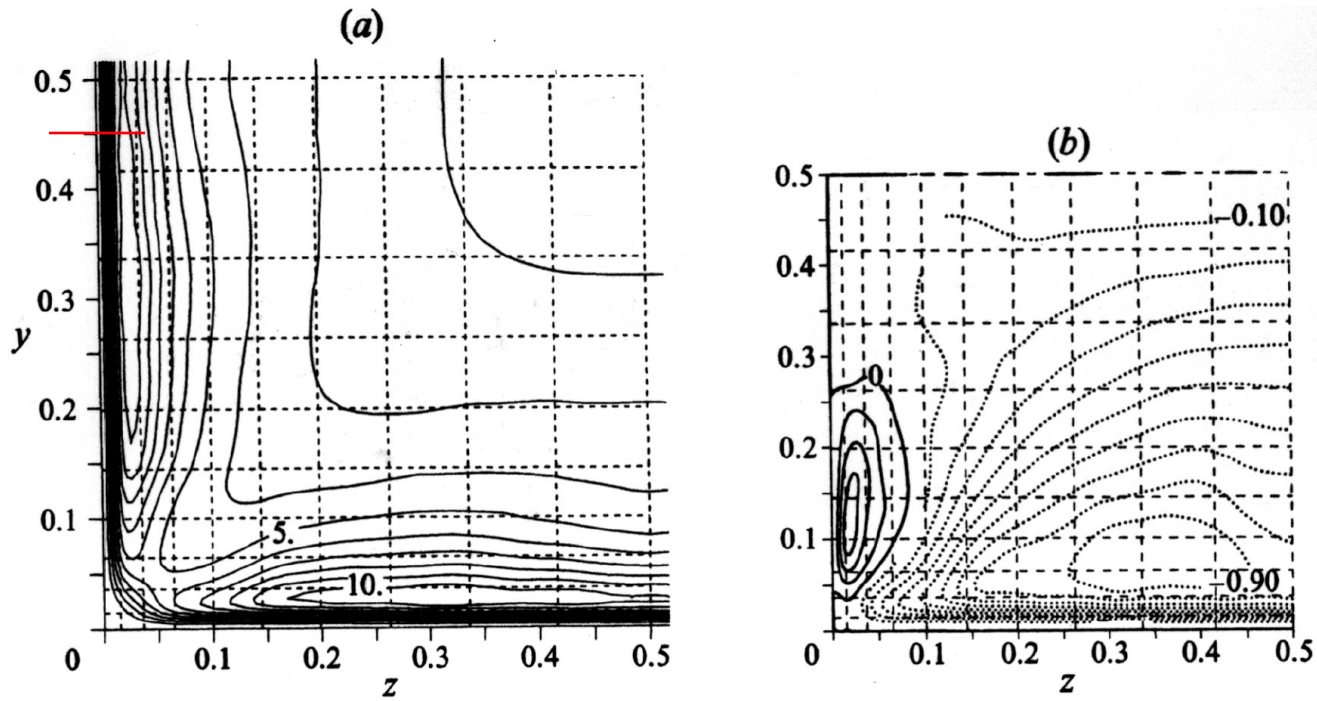


FIGURE 10. Primary Reynolds stress contours in the lower left quadrant of the transverse plane from Run B; every fifth grid-line is dashed: (a)  $\overline{u'^2}$ -contours, increment = 1.0; (b)  $\overline{u'v'}$ -contours, increment = 0.1.