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Transition to turbulence at the bottom of a solitary wave

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Motivation	Transition	Boundary layer	Stability	Results	Conclusions
Background					

Far from the coast the influence of surface waves on the bottom layer is insignificant. As the waves move closer to the coast the shear stress in the boundary layer increases and destabilizes the upper layers of sediment. Even closer to the coast the boundary layer changes from laminar to turbulent and sediment transport increases.

For relatively small water depths, sea waves can be modeled as eg. solitary waves.

Movie 1		

Exp. Sumer et al. (2010), JFM

The main differences between laminar and turbulent flow when it comes to sediment transport are

- Laminar flow : forces act locally on the sediment and the grain "diameter" becomes the important length scale
- Turbulent flow : large vortices "picks" up sediment, mixing, transport

It is therefore of importance to understand in what circumstances (parametrically) the flow transitions

Motivation	Transition	Boundary layer	Stability	Results	Conclusions





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Where do we start ?

Motivation	Transition	Boundary layer	Stability	Results	Conclusions





Where do we start ? Classical Modal analysis

Motivation Trans	tion Boundary lag	yer Stability	Results	Conclusions





Where do we start ? Classical Modal analysis Compare with DNS & Exp.

Motivation	Transition	Boundary layer	Stability	Results	Conclusions
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Definition of the basic flow : surface



No U Assume
$$\begin{split} H &= H^*/h^* << 1 \\ \mu &= h^*/L^* << 1 \mbox{ (Boussinesq)} \end{split}$$

with $H \sim \mu^2$, neglecting the wave damping and H^2 terms one obtains (Grimshaw, 1971) the free surface elevation and wave propagation velocity as

$$\eta^{*}(X_{1}^{*}, t^{*}) = H^{*} \operatorname{sech}^{2}\left(\sqrt{\frac{3H}{4}}\zeta\right)$$
$$V_{1}^{*}(X_{1}^{*}, t^{*}) = H\sqrt{g^{*}h^{*}}\operatorname{sech}^{2}\left(\sqrt{\frac{3H}{4}}\zeta\right)$$

where

$$\begin{aligned} \zeta &= (X_1^* - \sqrt{g^* h^* t^*})/h^* = X_1 - t \\ \text{ote that:} \\ {}^*_{ref} &= H\sqrt{g^* h^*} \qquad L^*_{ref} = H^* \quad \text{and} \quad Re = U^*_{ref} L^*_{ref} / \nu^* = H\sqrt{g^* h^*} H^* / \nu^* \sim (H/\delta) \end{aligned}$$

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Motivation	Transition	Boundary layer	Stability	Results	Conclusions

Definition of the basic flow : bottom boundary layer



$$Re_{\delta} = H\sqrt{g^*h^*}\delta^*/\nu^* = \sqrt{Re}$$

The upper (air) boundary layer is neglected (τ_{xy} small)

In the bottom boundary layer viscous and inertial effects should balance

$$\frac{\partial}{\partial t^*} \sim \sqrt{g^* h^*} / h^*, \quad \nu^* \frac{\partial^2}{\partial X_2^{*2}} \sim \nu^* / \delta^{*2}$$
$$\rightarrow \delta^* \sim \sqrt{\nu^* h^* / \sqrt{g^* h^*}}$$

Here : $\delta^*/h^* <<$ 1, consequently we can use

Boundary Layer Approximation

 v_{b2}^* is negligible (continuity equation) $\partial p^* / \partial X_2^* = 0$ (y momentum equation) v_{b1}^* is then obtain by solving

$$\frac{\partial v_{b1}^*}{\partial t^*} = \left. \frac{\partial V_1^*}{\partial t^*} \right|_{X_2=0} + \nu^* \frac{\partial^2 v_{b1}^*}{\partial X_2^{*2}}$$

o.c.: $v_{b1}^* = 0$ at $X_2^* = 0$ and $\left. \frac{\partial v_{b1}^*}{\partial X_2^*} \to 0$ as $\to \infty$

Motivation	Transition	Boundary layer	Stability	Results	Conclusions			
Definition of the basic flow : solution								

Following Mei, "The applied dynamics of ocean surface waves" (1989), the solution can be written as

$$v_{b1}(X_2,\zeta) = \operatorname{sech}^2\left(\sqrt{\frac{3H}{4}}\zeta\right) - \frac{2}{\sqrt{\pi}}\int_0^\infty \operatorname{sech}^2\left[\sqrt{\frac{3H}{4}}\left(\frac{1}{2}\frac{X_2^2}{\xi^2} + \zeta\right)\right]e^{-\xi^2}d\xi, \quad \text{with} \quad \zeta = X_1 - t$$

Case : Sumer et al. (2010), H = 0.12, $\delta = 0.0005$



Motivation	Transition	Boundary layer	Stability	Results	Conclusions
Linear sta	ability equati	ons			

$$(v_1, v_2, p) = (v_{b1}, 0, p_b) + \epsilon(v_{p1}, v_{p2}, p_p)$$
 where $\epsilon << 1$,

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and $U_{ref}^* = H\sqrt{g^*h^*}$, $L_{ref}^* = \delta^*$, $t_{ref}^* = L_{ref}^*/U_{ref}^*$, $p_{ref}^* = \rho^*Hg^*\delta^*$.

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Linear et	ability any at				
Linear su	ability equati	ons			

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We introduce the stream function: $v_{p1} = \partial \psi / \partial x_2$ and $v_{p2} = -\partial \psi / \partial x_1$.

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If we consider $H^* >> \delta^* \Rightarrow H/\delta >> 1$, then the perturbation amplitude grows on a time scale much faster that the basic flow.

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Linear stah	ility equati	ons			

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The modal form of the stream function can therefore be written

$$\psi(\mathbf{x}_1, \mathbf{x}_2, t) = f(\mathbf{x}_2, t) \exp\left[i\alpha\left(\mathbf{x}_1 - \frac{H}{\delta}\int c(\tau)d\tau\right)\right],$$

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and the governing equation at order $\boldsymbol{\epsilon}$ becomes

$$[v_{b1}(x_2,t) - c(t)]\Delta f(x_2,t) - \frac{\partial^2 v_{b1}}{\partial x_2^2} f(x_2,t) = \frac{1}{2i\alpha(H/\delta)} \Delta^2 f(x_2,t),$$

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This is an eigenvalue problem for the complex valued variable c(t).

The so called dispersion relation can be written $c = c(H, \delta, \alpha, \zeta) = c_r + ic_i$.

 c_r : phase speed, c_i : growth rate, $c_i > 0 \Rightarrow$ unstable solution.

Motivation	Transition	Boundary layer	Stability	Results	Conclusions
Results					

The results are presented in the following way

- Experiments by Sumer et al. (2010)*
 - U-shaped water tunnel excited by piston mechanism
 - $L \times H \times B = 10 \times 0.29 \times 0.39 m^3$
 - Flow visualization with color CCD camera (25 frames/second)
 - shear stress (hot film probe) and free stream velocity (Laser doppler anemometer, LDA) measurements
- Linear Stability Analysis : critical conditions (ζ , α)
- Comparison with Direct Numerical Simulation

*Sumer et al. (2010), "Coherent structures in wave boundary layers. Part 2. Solitary motion", Journal of Fluid Mechanics, 646, 207-231

Motivation Transition Boundary layer Stability Results Conclusions

Video (plan view) from Sumer et al. (2010)

Video from experiments by sumer et al. (2010)^{*} where H = 0.12, $\delta = 0.0005$, flow from left to right. The video shows the vortex tubes in plan view.



*Sumer et al. (2010), "Coherent structures in wave boundary layers. Part 2. Solitary motion", Journal of Fluid Mechanics, **646**, 207-231

Motivation	Transition	Boundary layer	Stability	Results	Conclusions
Video (sid	de view) fro	m Sumer et al.	(2010)		

Video from experiments by sumer et al. (2010)* where H = 0.11, $\delta = 0.00054$, flow from left to right. The video shows the vortex tubes in side view.

Movie 2

*Sumer et al. (2010), "Coherent structures in wave boundary layers. Part 2. Solitary motion", Journal of Fluid Mechanics, **646**, 207-231

Motivation	Transition	Boundary layer	Stability	Results	Conclusions
Example	result from L	ST: eigenfunct	tions		

Case : Sumer et al. (2010), $H = 0.12, \ \delta = 0.0005, \ \alpha = 0.2$

Movie 1

Motivation	Transition	Boundary layer	Stability	Results	Conclusions
Linear st	ability results				

A worst case scenario can be assumed which requires to "scan" the whole parameter space, $c = c(H, \delta, \alpha, \zeta)$. In such a way critical conditions can be established as shown in the figure. Here it is shown that the instability occurs for $\zeta > 0$ which means the deceleration phase.

In this case $\zeta_c = 1.0$ and the corresponding wave number $\alpha_c = 0.2$.



FIGURE 2. (a) Growth rate $c_i \ (\Delta c_i = 0.005, \text{ only positive values are plotted})$; and (b) phase velocity $c_r \ (\Delta c_r = 0.02, \text{ continuous lines}, c_r \ge 0)$; broken lines $c_r < 0$). plotted versus ζ and α for H = 0.12 and $\delta = 5.0 \times 10^{-4}$.

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Motivation	Transition	Boundary layer	Stability	Results	Conclusions
Comparison	with DNS				

In Direct Numerical Simulations (DNS) the flow is computed without any approximations. It is therefore a "numerical experiment" to compare the Linear Stability (LST) results with. Two different DNS computations have been performed.

- Given initial condition of the perturbations
- Model of distributed wall roughness during the whole wave cycle

This gives different Receptivity scenarios and it is shown that the latter agrees better with LST (worst case scenario).



FIGURE 3. (a) Dimensionless kinetic energy per unit area K of the perturbations of the laminar boundary layer under a solitary wave, computed using the numerical approach of Vittori & Blondeaux (2008) for H = 0.20 and $\delta = 8 \times 10^{-4}$. The broken line is the value of K obtained introducing a perturbation of the laminar flow at the beginning of the numerical simulation and considering a perfectly plane wall, the continuous line is the value obtained introducing wall imperfections and considering vanishing initial condition. (b) Growth rate c_i plotted versus ζ for H = 0.20, $\delta = 8 \times 10^{-4}$ and three different values of α .

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Motivation	Transition	Boundary layer	Stability	Results	Conclusions
Summary	of results .	comparison bet	ween IST a	nd experim	ents

Summary of experiments by Sumer et al. (2010) in comparison with Linear stability results. A reasonable agreement is found regarding the critical wave number α_c , while the critical time (LST) is under estimated.

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Exp. no :	H	δ	α_{c} LST	$\alpha_c \exp$	ζ_c LST	$\zeta_c \exp$
1	0.12	0.0005	0.2	0.21-0.3	1.01	3.18
2	0.108	0.00054	0.2	0.23-0.3	1.16	4.77
3	0.199	0.00043	0.21	0.23-0.27	0.53	2.23
4	0.096	0.0006	0.205	0.19-0.26	1.39	4.81

Motivation	Transition	Boundary layer	Stability	Results	Conclusions
Conclusions					

- The solitary wave boundary layer is unstable if the height H exceeds a certain threshold, for a given boundary layer thickness δ .
- The instability sets in during the deceleration phase (for the parameters investigated).
- The critical wave length found by LST is similar to the distance between the vortex tubes found in the experiments by Sumer et al. (2010)
- The threshold wave height is under estimated by LST
- The discrepancy between DNS and LST might be explained considering different receptivity scenarios.

The work has been accepted for publication in the Journal of Fluid Mechanics